



**SUBJECT : CHARACTERISTIC AND PROTECTION SPEEDS**

**INTRODUCTION**

The different speeds displayed to the crew on the main cockpit interfaces : PFD, MCDU, ND are computed by the FACs, the FMGCs and the ADIRS.

PFD	MCDU PERF PAGE
<p style="text-align: center;">FAC COMPUTATION</p> <p>Computed on current aircraft status and configuration.</p>	<p style="text-align: center;">FMGC COMPUTATION</p> <p>Computed for take off, go around and landing.</p>
<p>VLS F S "O" Green Dot V<math>\alpha</math>prot V<math>\alpha</math>max Vsw (stall warning speed)</p>	<p>VLS of the selected landing configuration. F S "O" Green Dot</p>

Each FAC computes its own speeds which are displayed on the relevant PFD.

- FAC 1 on side 1
- FAC 2 on side 2

Each FMGC computes its own speeds displayed on the relevant MCDU :

- FMGC 1 on side 1
- FMGC 2 on side 2

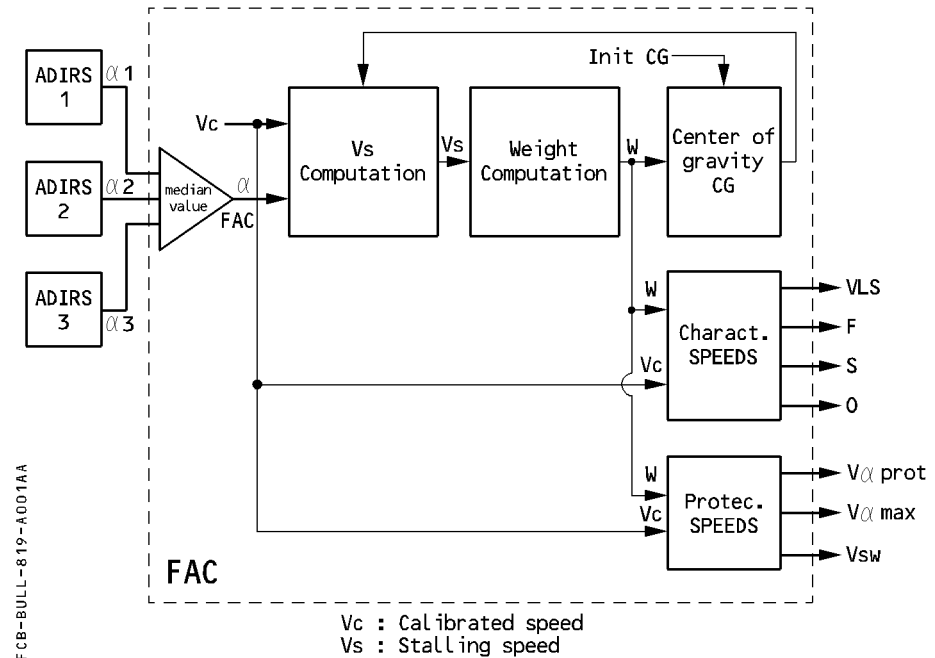
The algorithms used to compute the characteristic speeds are the same in both FAC and FMGC but as the inputs are different, the resulting values may differ.

## CHARACTERISTICS SPEEDS COMPUTED BY THE FAC

The FAC computes its characteristic speeds with 2 main inputs from ADIRS (Angle of Attack ( $\alpha$ ) and calibrated airspeed ( $V_c$ )). It also uses THS position, SFCC data and FADEC data.

From these inputs, the FAC computes a stall speed  $V_s$  which is used to determine the aircraft weight.

The following sketch gives the basic architecture for FAC speed computation.

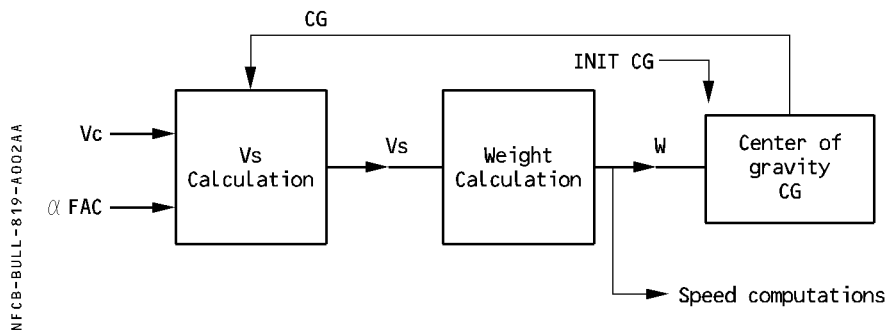


## AOA DETERMINATION

The angle of attack value used to compute the characteristic speeds is the mean value of the 3 AOAs (Vote).

Accuracy of the AOAs is a paramount factor in the weight calculation. 0.3 degree of error in the AOA results in a 3 ton error in weight.

## WEIGHT COMPUTATION



The weight is computed provided the following conditions are met.

- Aircraft altitude below 14600 ft and speed ( $V_c$ ) below 240 kt
- Bank angle less than  $5^\circ$
- Speedbrakes retracted
- No dynamic maneuver (vertical load factor lower than 1.07 g)
- No change of aircraft configuration and not in conf full.

When one of these conditions is not met, the last calculated weight value is considered and updated for the fuel consumption based on actual engine N1.

## CHARACTERISTIC SPEEDS COMPUTATION

### **A320**

VLS is computed from Weight and  $V_c$  and corrected for the current CG.

- If the current CG is forward of 15 %, 15 % CG is used to compute the speeds.
  - If the current CG is between 15 % and 25 %, the speeds are computed using an interpolation between 15 % and 25 % CG.
  - If the current CG is aft of 25 %, 25 % CG is used for speed computation.
- F, S, and Green dot are independent of CG.

## A319-A321

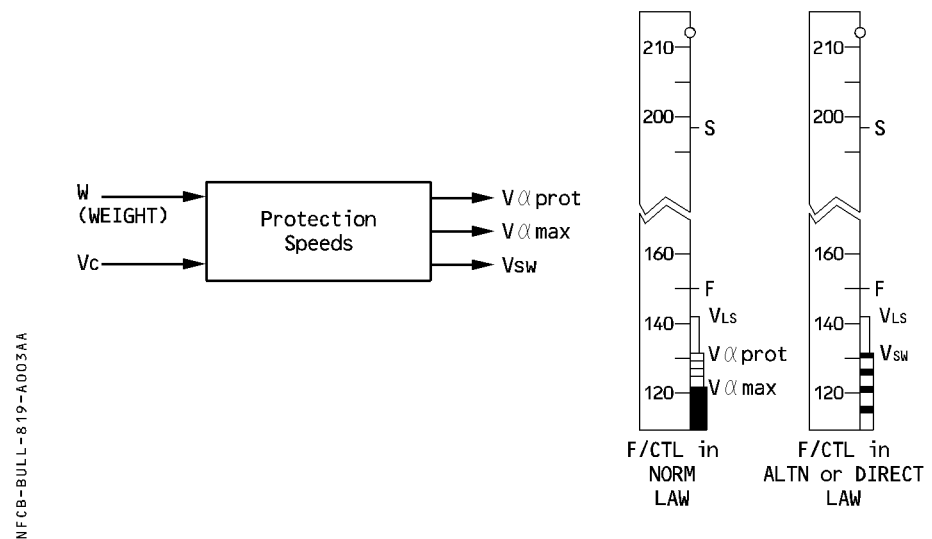
VLS, F, S and Green Dot are computed for a forward CG. No CG correction is applied for A319/A320 VLS as it has a negligible effect.

### PROTECTION SPEEDS CALCULATED BY THE FAC

$V_{\alpha prot}$  and  $V_{\alpha max}$  are displayed in normal law.

The FAC does not trigger alpha prot and alpha max protection.  
(The alpha prot and alpha max protection are activated by the ELAC).

Vsw, the stall warning speed is computed by the FAC in ALTN or DIRECT law. At Vsw speed, an audio warning (crickets – STALL synthetic voice) is triggered.



### TOLERANCE OF FAC COMPUTED SPEEDS

Due to the data accuracy used to compute the characteristic speeds, and specifically the AOA accuracy, the precision of the computation is specified to be within 2.5 %.

During acceptance flight, the tolerances are as following :

Clean aircraft	Green Dot	$\pm 5$ kt
	VLS	$\pm 4$ kt
	$V_{\alpha prot}$	$\pm 5$ kt
	$V_{\alpha max}$	$\pm 5$ kt
Conf full	VLS	$\pm 3$ kt
	$V_{\alpha prot}$	$\pm 5$ kt
	$V_{\alpha max}$	$\pm 5$ kt

## CHARACTERISTICS SPEEDS COMPUTED BY THE FMGC

Characteristic speeds computed by the FMGC are based on a predicted GW, CG (and selected configuration for landing) at a given time at landing for example.

GW and CG values are computed from entered ZFW and ZFWCG corrected for the predicted FOB and CG variation.

When the Approach phase is activated, the characteristic speeds are recomputed using the actual weight and CG.

The performance model used to compute the characteristic speeds, is accurate enough to provide speed errors of less than  $\pm 2$  kt from the certified speeds.

NFCB-BULL-819-A004AA

APPR			FINAL			TAKE OFF					
1L	DEST	QNH	FLP	RETR	F=163	VOR33R	V1	FLP	RETR	RWY	23
2L	TEMP	[ ]°	S=196	SLT	RETR	MDA	145	S=196	[M]	900	2R
3L	MAG WING	[ ]°/[ ]	0=236	CLEAN			V2	CLEAN	FLAPS/THS		3R
4L	TRANS ALT			LDG CONF			4800	TRANS ALT	FLEX TO TEMP	45°	4R
5L	VAPP	VLS	127	FULL			THR RED/ACC	ENG OUT ACC		2865	5R
6L	PREV			NEXT			3000/4305	NEXT			6R
	<PHASE			PHASE>				PHASE>			

## THE MOST FREQUENT QUESTIONS ON SPEED COMPUTATION

- Why are the characteristic speeds computed by the FAC subject to inaccuracy greater than FMGC computation ?

Answer :

The precision of the AOA measurement is usually the cause of speed differences. An error of 0.3 ° in AOA measurement causes a weight inaccuracy close to 3 tons.

- Is the FMGC computation more accurate than the FAC computation ?

Answer :

Algorithms are the same but the initial data are different.

The FAC computes current dynamic speeds.

The FMGS computes characteristic speeds for given phases (and configuration for landing).

Usually, the FMGC characteristic speeds for landing are more accurate due to the tolerance of FAC inputs, if the ZFW was correct initially.

Note : To determine the GW at landing, the FMGC uses the ZFW entered by the crew and adds the fuel on board.

A significant difference between PFD and MCDU characteristic speeds may also indicate an error in the ZFW as entered by the crew.

– **Why are there two characteristic speed calculations ?**

Answer :

- The computation done by the FAC is independent of any manually entered data and provides permanent speed values displayed on the PFD.
- During approach, the comparison of characteristic speeds allows the crew to detect any speed discrepancy which may affect approach and final phases ;

– **When a difference exists between computed speeds from FAC and FMGC, what are the best speeds to be relied on ?**

Answer :

Whenever differences are observed, Airbus recommends to rely on QRH values.  
Refer to FCOM 4.06.20 p 7.